Reinforcement Learning for Orientation Estimation Using Inertial Sensors with Performance Guarantee

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Abstract—This paper presents a deep reinforcement learning (DRL) algorithm for orientation estimation using inertial sensors combined with magnetometer. The Lyapunov's method in control theory is employed to prove the convergence of orientation estimation errors. Based on the theoretical results, the estimator gains and a Lyapunov function are parametrised by deep neural networks and learned from samples. The DRL estimator is compared with three well-known orientation estimation methods on both numerical simulations and real dataset collected from commercially available sensors. The results show that the proposed algorithm is superior for arbitrary estimation initialisation and can adapt to very large angular velocities for which other algorithms can be hardly applicable. To the best of our knowledge, this is the first DRL-based orientation estimation method with estimation error boundedness guarantee.

I. INTRODUCTION

Orientation estimation is important in robotics, navigation, control, human motion analysis [1], [2], [3]. Recently, orientation estimation has been greatly advanced by the development of accurate sensors. Multiple sensors are usually combined to estimate the orientation, i.e., sensor fusion. Depending on the availability of sensors and applications, various sensor fusion techniques have been proposed, e.g., the inertial measurement units (IMU) and magnetometer [4], [5], [6], the magnetometer and camera [7], and the IMU and visual sensor [8], [9], etc. In this paper, we focus on orientation estimate using the inertial sensors and magnetometer.

The estimation algorithms can be summarised into three categories: (1) Bayesian estimation, (2) optimisation and (3) deep learning. In Bayesian estimation, the well-known extended Kalman filter (EKF) and the unscented Kalman filter (UKF), were used to estimate the orientation [4], [5], [10]. The key idea is to approximate the orientation states by a Gaussian distribution based on the linearisation technique and the deterministic sampling technique, respectively. Furthermore, the complementary filter was developed based on the EKF, which exploits the complementary characteristics of gyroscopes and that of accelerometer and magnetometer at different time scales [6]. In optimisation, the orientation estimation is obtained based on gradient-based optimisation algorithms [11], [12]. Until recently, the deep learning was introduced to estimate the orientation [13], in which a deep neural network is trained to mimic the noise distribution of gyroscopes

such that accurate orientation estimates can be obtained by open-loop integration of the noise-free gyro measurements. These algorithms showed superior estimation performance empirically. However, the performance can not be theoretically guaranteed, i.e, the orientation estimate error never diverge. In this paper, we will employ Lyapunov's method in control theory to prove the estimation error boundedness guarantee using samples. Based on the theoretical result, we will develop a DRL-based algorithm to learn the estimator from samples.

Reinforcement learning (RL) was first applied for state estimation in [14]. Motivated by this work, we plan to develop a RL algorithm to learn the estimator gain using samples while the orientation estimator keeps the EKF structure. More specifically, the estimator gain will be approximated by a deep neural network (DNN) as a function of the sequence of estimate errors. Different from other popular RL algorithms [15], [16], [17], the value function will be treated as a Lyapunov function to guarantee the estimation performance. Lyapunov's method has been widely used as a basic tool for stability analysis in control theory [18]. To analyse the stability, the key is to find a scalar "energylike" Lyapunov function for the considered system such that the derivative/difference of Lyapunov function along the state trajectory is semi-negative definite. Nonetheless, the construction/learning of the Lyapunov function in modelfree reinforcement learning is not trivial. [19] proposes a straightforward approach to construct the Lyapunov functions for nonlinear systems using DNNs. Recently, the asymptotic stability in model-free RL is given for robotic control tasks in [20]. Inspired by the works [19], [20], we will also parametrise the Lyapunov function as a DNN and learn the parameters from samples. Thereafter, a new DRL algorithm based on soft actor-critic algorithm [17] that incorporates the Lyapunov boundedness condition in the objective function to be optimised is proposed. By using the learned estimator gain, the estimate error of the orientation estimator is guaranteed to be bounded all the time.

In summary, we combine Lyapunov's method and DRL to design a state estimator with estimation error boundedness guarantee for orientation estimation. The main contribution of this paper is threefold:

- To the best of our knowledge, this is the first DRL-based orientation estimation method using inertial sensors combined with magnetometer;
- 2) Estimation error boundedness guarantee is proved using Lyapunov's method in control theory;
- 3) The proposed algorithm is superior for arbitrary estimation initialisation and can adapt to very large angular

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velocities for which other algorithms, such as the EKF, UKF and complementary filter algorithms, can be hardly applicable.

The rest of the paper is organised as follows. In Section II, the orientation estimation problem is formulated followed by the preliminaries of Markov decision process. In Section III, the theoretical result on estimation error boundedness guarantee is proved. In Section IV, a DRL algorithm based on Soft Actor Critic (SAC) combined with theoretical results is proposed to learn the estimator gain. In Section V, our method is compared with the EKF, UKF and complementary filter algorithms in simulations and experiments. Conclusion is given in Section VI.

II. PROBLEM FORMULATION

In this paper, we use the inertial sensors (3D accelerometers and 3D gyroscopes) combined with the magnetometer to estimate the orientation. As in [3], the system dynamics is standard and our goal is to design the estimator gain. Different from the classic nonlinear filtering techniques based on linearisation, we will show that the estimator gain computation can be formulated as a RL problem.

A. System dynamics and state estimator

The orientation dynamics is typically given as:

$$q_{t+1}^{\mathsf{nb}} = q_t^{\mathsf{nb}} \odot \exp_q \left(\frac{T}{2} (y_{\omega,t} - e_{\omega,t}) \right), \tag{1}$$

where $q_t^{nb} \in \mathbb{R}^4$ is the unit quaternion for the orientation of the body frame with respect to the navigation frame at time instant t, T is the sampling time, and $y_{\omega,t}$ is the gyroscope measurement. The noise of the gyroscope is a zero-mean Gaussian noise $e_{\omega,t} \sim \mathcal{N}(0, \Sigma_{\omega})$ where Σ_{ω} is the covariance of the noise.

Assuming that the linear acceleration is approximately zero, the measurement equations are given as follows:

$$y_{\mathbf{a},t} = -R_t^{\mathbf{b}\mathbf{n}}g^{\mathbf{n}} + e_{\mathbf{a},t},\tag{2a}$$

$$y_{\mathrm{m},t} = R_t^{\mathrm{bn}} m^{\mathrm{n}} + e_{\mathrm{m},t},\tag{2b}$$

where $y_{a,t}, y_{m,t} \in \mathbb{R}^3$ are accelerometer and magnetometer measurements at time instant t, respectively, R_t^{bn} is the rotation matrix from the navigation frame to the body frame at time instant t, g^n, m^n denote the local earth gravity vector and the local earth magnetic field vector, respectively. The noises $e_{a,t} \sim \mathcal{N}(0, \Sigma_a)$, and $e_{m,t} \sim \mathcal{N}(0, \Sigma_m)$ with $\Sigma_m = \sigma_m^2 \mathcal{I}_3$ and $\Sigma_a = \sigma_a^2 \mathcal{I}_3$.

To estimate q_{t+1}^{nb} , the following estimator in terms of the orientation deviation is often proposed [3], [21]:

$$\hat{q}_{t+1|t}^{\text{nb}} = \hat{q}_{t|t}^{\text{nb}} \odot \exp_q\left(\frac{T}{2}y_{\omega,t}\right),\tag{3a}$$

$$\hat{\eta}_{t+1} = K_{t+1}(y_{t+1} - \hat{y}_{t+1|t}), \qquad (3b)$$

$$\hat{q}_{t+1|t+1}^{\text{nb}} = \exp_q\left(\hat{\eta}_{t+1}\right) \odot \hat{q}_{t+1|t}^{\text{nb}}$$
 (3c)

with

$$y_t = \begin{pmatrix} y_{\mathbf{a},t} \\ y_{\mathbf{m},t} \end{pmatrix}, \quad \hat{y}_{t+1|t} = \begin{pmatrix} -R \left\{ \begin{array}{c} \hat{q}_{t|t}^{\mathsf{nb}} \odot \exp_q(\frac{T}{2}y_{\omega,t}) \right\}^\top g^n \\ R \left\{ \hat{q}_{t|t}^{\mathsf{nb}} \odot \exp_q(\frac{T}{2}y_{\omega,t}) \right\}^\top m^n \end{pmatrix},$$

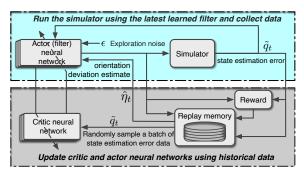


Fig. 1. Offline RL training process of orientation state estimator

where $\hat{q}_{t+1|t}^{nb}$ is the linearisation point parametrised in terms of quaternions, $\hat{\eta}_{t+1}^{n}$ is the state estimate of the orientation deviation, and $R\{\}$ denotes the matrix formula of translation from quaternion to rotation. The goal is to obtain K_{t+1} , i.e., the estimator gain at time instant t + 1, which will be described in detail in the next subsection.

Define the orientation error

$$\tilde{q}_t \triangleq q_t^{\rm nb} \odot \left(\hat{q}_{t|t}^{\rm nb} \right)^{\rm c}, \tag{4}$$

or equivalently $q_t^{\rm nb} = \tilde{q}_t \odot \hat{q}_{t|t}^{\rm nb}$. From (1), (2) and (3), we have

$$\widetilde{q}_{t+1} = q_{t+1}^{\text{nb}} \odot \left(\widehat{q}_{t+1|t+1}^{\text{nb}} \right)^{\text{c}} \\
= \left(\left(\widetilde{q}_t \odot \widehat{q}_{t|t}^{\text{nb}} \right) \odot \exp_q \left(\frac{T}{2} (y_{\omega,t} - e_{\omega,t}) \right) \right) \odot \\
\left(\left(\exp_q \left(\frac{1}{2} K_{t+1} (y_{t+1} - \widehat{y}_{t+1|t}) \right) \odot \widehat{q}_{t|t}^{\text{nb}} \odot \exp_q \left(\frac{T}{2} y_{\omega,t} \right) \right) \right)^{\text{c}} \\$$
(5)

Furthermore, to escape the unit determinant condition of the quaternion representation of rotation, the logarithm map of the quaternion is used [22]:

$$[\eta_{t+1}]_{\times} = \log(\tilde{q}_{t+1}) \tag{6}$$

where η_{t+1} is the orientation deviation and the skew operator $[\cdot]_{\times}$ produces the cross-product matrix.

B. Estimate Error Dynamics as Markov Decision Process

By combining (5) and (6), the dynamics of estimate error can be viewed as a Markov decision process (MDP) which is defined as a tuple $\langle S, A, P, C, \gamma \rangle$:

$$\tilde{q}_{t+1} \sim \mathcal{P}\left(\tilde{q}_{t+1} | \tilde{q}_t, K_{t+1}\right), \forall t \in \mathbb{Z}_+,\tag{7}$$

where the estimate error $\tilde{q}_t \in S$ is the state, the estimator gain $K_{t+1} \in A$ is the action sampled from a stochastic policy. Considering that the ground truth q_t is known during training phase, the mapping between \tilde{q}_t and $\hat{\eta}_t$ is bijective to some extent according to (3c) and (4). For convenience, in our implementation of the algorithm, we treat $\pi(K_{t+1}|\hat{\eta}_t)$ and $\pi(K_{t+1}|\tilde{q}_t)$ equivalently.

The state dynamics can be characterised by the transition probability function $\mathcal{P}(\tilde{q}_{t+1}|\tilde{q}_t, K_{t+1})$. RL algorithms can be used to find the policy π , given a cost function¹

¹We will use cost instead of reward in this paper which is often used in control literature. Maximisation in RL setup will be minimisation instead.

 $C(\tilde{q}_t, K_{t+1}) \in \mathcal{C}$ that measures the goodness of a stateaction pair. In state estimation, it is often desired that the estimate error \tilde{q}_t converges exponentially to a finite bound in mean square. As such, the cost function is selected as $C(\tilde{q}_t, K_{t+1}) = \mathbb{E}_{P(\cdot|\tilde{q}_t, K_{t+1})}[\|\tilde{q}_{t+1}\|^2]$, and the return is the sum of discounted cost $\sum_{\tau=t}^{\infty} \gamma^{\tau-t} C(\tilde{q}_t, K_{t+1})$ with the discount factor $\gamma \in [0, 1)$.

Definition 1: [23] The estimate error \tilde{q}_t in the MDP (7) is said to be exponentially bounded in mean square if $\exists \eta > 0$ and $0 < \varphi < 1$, such that

$$\mathbb{E}[\|\tilde{q}_t\|^2] \le \eta \mathbb{E}[\|\tilde{q}_0\|^2]\varphi^t + p, \tag{8}$$

holds at all the time instants $t \ge 0$, where p is a positive constant number.

In this paper, our aim is to learn the estimator gain $K_{t+1} = \pi(\hat{\eta}_t^n)$ in (3) as a policy using RL, as showed in Fig.1. We expect that the mean square of the estimate error of \tilde{q}_t in (7) is guaranteed to converge exponentially to a positive bound. Different from the EKF where K_{t+1} is computed using the linearisation approximation, in this paper K_{t+1} is approximated by a deep neural network $\pi(\cdot)$.

III. ESTIMATION ERROR BOUNDEDNESS GUARANTEE

In this section, we propose the main theorem on boundedness of the estimate error. Before proceeding further, some notations are to be defined. $\rho(\tilde{q}_0)$ denotes the distribution of the starting state estimate error \tilde{q}_0 . The state distribution of state estimate error at a certain instant t as $P(\tilde{q}_t|\rho, \pi, t)$ is defined in an iterative way: $P(\tilde{q}_{t+1} = s'|\rho, \pi, t+1) = \int_S P(\tilde{q}_t = s|\rho, \pi, t)P_{\pi}(s'|s)ds$. The following assumption that will be used in proving the theorem is presented as follows:

Assumption 1: The Markov chain in (7) induced by a policy π is ergodic with a unique distribution probability. That is, $\exists p_{\pi}(s)$, such that

$$p_{\pi}(s) = \lim_{t \to \infty} P(\tilde{q}_t = s | \rho, \pi, t) \tag{9}$$

Theorem 1: The error dynamics (7) is exponentially bounded in mean square if there exists a Lyapunov function $L(\tilde{q}_t): S \to R^+$ and positive constants α_1, α_2 and δ such that

$$\alpha_1 \mathbb{E}_{\pi}[\|\tilde{q}_t\|^2] - \delta \le L(\tilde{q}_t) \le \alpha_1 \mathbb{E}_{\pi}[\|\tilde{q}_t\|^2]$$
(10)

and

$$\lim_{N \to +\infty} [\ln(\mathbb{E}_{\tilde{q}_t \sim \mu_N}(\mathbb{E}_{\tilde{q}_{t+1} \sim P_{\pi}}L(\tilde{q}_{t+1}))) -\mathbb{E}_{\tilde{q}_t \sim \mu_N}\ln(L(\tilde{q}_t))] \le -\alpha_2$$
(11)

where

$$\mu_N(s) \triangleq \frac{1}{N} \sum_{t=0}^{N-1} P(\tilde{q}_t = s | \rho, \pi, t)$$
(12)

Proof: We have

$$\ln(\mathbb{E}_{\tilde{q}_{t}\sim\mu_{N}}(\mathbb{E}_{\tilde{q}_{t+1}\sim P_{\pi}}L(\tilde{q}_{t+1})))$$

$$=\ln(\int_{S}\frac{1}{N}\sum_{t=0}^{N-1}P(\tilde{q}_{t}=s|\rho,\pi,t)\int_{S}P_{\pi}(s'|s)L(s')\,ds'\,ds)$$

$$=\ln(\int_{S}(\int_{S}\frac{1}{N}\sum_{t=0}^{N-1}P(\tilde{q}_{t}=s|\rho,\pi,t)P_{\pi}(s'|s)\,ds)L(s')\,ds')$$

$$=\ln((\frac{1}{N}\sum_{t=0}^{N-1}\int_{S}P(\tilde{q}_{t+1}=s'|\rho,\pi,t+1))L(s')\,ds')$$

$$\geq\frac{1}{N}\sum_{t=0}^{N-1}\ln((\int_{S}P(\tilde{q}_{t+1}=s'|\rho,\pi,t+1))L(s')\,ds')$$
(13)

where the last inequality follows from the fact that $\ln(x)$ is a concave function on R^+ . Similarly, noting that $-\ln(x)$ is a convex function we have

$$-\mathbb{E}_{\tilde{q}_{t}\sim\mu_{N}}\ln L(\tilde{q}_{t})$$

$$=-\int_{S}\frac{1}{N}\sum_{t=0}^{N-1}P(\tilde{q}_{t}=s|\rho,\pi,t)\ln(L(s))\,ds$$

$$=\frac{1}{N}\sum_{t=0}^{N-1}\int_{S}P(\tilde{q}_{t}=s|\rho,\pi,t)(-\ln L(s))\,ds$$

$$\geq\frac{1}{N}\sum_{t=0}^{N-1}-\ln(\int_{S}P(\tilde{q}_{t}=s|\rho,\pi,t)L(s)\,ds)$$
(14)

It follows from the above two inequalities that

$$\ln(\mathbb{E}_{\tilde{q}_{t}\sim\mu_{N}}(\mathbb{E}_{\tilde{q}_{t+1}\sim P_{\pi}}L(\tilde{q}_{t+1})) - \mathbb{E}_{\tilde{q}_{t}\sim\mu_{N}}\ln L(\tilde{q}_{t})) \\
\geq \frac{1}{N} \sum_{t=0}^{N-1} \ln \frac{\int_{S} P(\tilde{q}_{t+1} = s'|\rho, \pi, t+1)L(s') \, ds'}{\int_{S} P(\tilde{q}_{t} = s|\rho, \pi, t)L(s) \, ds}$$

$$\geq \frac{1}{N} \sum_{t=0}^{N-1} \ln \frac{\mathbb{E}_{\tilde{q}_{t+1}}L(\tilde{q}_{t+1})}{\mathbb{E}_{\tilde{q}_{t}}L(\tilde{q}_{t})}$$
(15)

Substituting the above into (11), we obtain

$$\lim_{N \to +\infty} \frac{1}{N} \sum_{t=0}^{N-1} \ln \frac{\mathbb{E}_{\tilde{q}_{t+1}} L(\tilde{q}_{t+1})}{\mathbb{E}_{\tilde{q}_t} L(\tilde{q}_t)} \le -\alpha_2$$
(16)

then

$$\lim_{N \to +\infty} \frac{1}{N} \ln \frac{\mathbb{E}_{\tilde{q}_N} L(\tilde{q}_N)}{\mathbb{E}_{\tilde{q}_0} L(\tilde{q}_0)} \le -\alpha_2 \tag{17}$$

It means that $\forall \epsilon > 0, \exists N_{\epsilon}, \frac{1}{N} \ln \frac{\mathbb{E}_{\tilde{q}_N} L(\tilde{q}_N)}{\mathbb{E}_{\tilde{q}_0} L(\tilde{q}_0)} < -\alpha_2 + \epsilon < 0$ holds when $N > N_{\epsilon}$, namely

$$\frac{\mathbb{E}_{\tilde{q}_N} L(\tilde{q}_N)}{\mathbb{E}_{\tilde{q}_0} L(\tilde{q}_0)} \le e^{N(-\alpha_2 + \epsilon)}, \forall N > N_{\epsilon}$$
(18)

So we get for sufficiently large $N > N_{\epsilon}$,

$$\mathbb{E}_{\tilde{q}_N \sim P(\tilde{q}_N | \rho, \pi, N)} L(\tilde{q}_N) \le e^{N(-\alpha_2 + \epsilon)} \mathbb{E}_{\tilde{q}_0 \sim \rho(\tilde{q}_0)} L(\tilde{q}_0)$$
(19)

By Equation (10) we have the following result

$$\mathbb{E}_{\tilde{q}_N \sim P(\tilde{q}_N | \rho, \pi, N)} \mathbb{E}_{\pi} \| \tilde{q}_N \|^2$$

$$\leq e^{N(-\alpha_2 + \epsilon)} \mathbb{E}_{\tilde{q}_0 \sim \rho(\tilde{q}_0)} \mathbb{E}_{\pi} \| \tilde{q}_0 \|^2 + \frac{\delta}{\alpha_1}$$
(20)

So far, it has been proved that the estimate error \tilde{q}_t in (7) is exponentially bounded according to Definition 1.

IV. LYAPUNOV-BASED REINFORCEMENT LEARNING ORIENTATION ESTIMATION ALGORITHM

In this section, we will combine Soft Actor-Critic (SAC) algorithm [17], one of the state-of-the-art RL algorithms, with the theoretical result in Section III to learn the gain policy K_{t+1} for the state estimator (3).

Considering MDP in (7), the orientation estimation problem can be viewed as a RL problem in which the policy is sought after by minimising the expected accumulated cost, i.e., the value function of the MDP. Here a stochastic policy is chosen as $\pi(K_{t+1} \mid \tilde{q}_t) \sim \mathcal{N}(K_{t+1}(\tilde{q}_t), \sigma)$ from which the gain K_{t+1} for a given state \tilde{q}_t is sampled [24]. The corresponding Q-function (a.k.a, state-action value function) is given as:

$$Q_{\pi}(\tilde{q}_t, K_{t+1}) = C_t(\tilde{q}_t, K_{t+1}) + \gamma \mathbb{E}[V_{\pi}(\tilde{q}_{t+1})]$$
(21)

To this end, K_{t+1} can be learned by many existing RL algorithms.

Motivated by the works in [20], [25], [26], [27], we propose to incorporate the theoretical result in Theorem 1 to formulate a constrained optimisation problem, based on SAC [17]. First of all, a Lyapunov candidate needs to be selected at the first instance. In the context of RL, a Lyapunov candidate will be parametrised/selected as the Q-function [19], [28]. In this paper, we choose $L(\tilde{q}_t)$ in (10) as:

$$L(\tilde{q}_t) = \mathbb{E}_{K_{t+1} \sim \pi} [L_c(\tilde{q}_t, K_{t+1})]$$
(22)

where $L_c(\tilde{q}_t, K_{t+1}) = Q(\tilde{q}_t, K_{t+1})$. Then the constrained optimisation problem is:

$$\min_{\pi} \quad Q_{\pi}(\tilde{q}_t, K_{t+1})$$
s.t. (10) and (11)

$$-\ln(\pi(K_{t+1} \mid \tilde{q}_t)) \geq \mathcal{H}_t$$
(23)

where $Q_{\pi}(\tilde{q}_t, K_{t+1})$ is defined in (21), the second constraint is the minimum entropy constraint used in the SAC to improve the exploration in the action space [29] where \mathcal{H}_t is the desired bound.

Denote the parametrised actor and critic as $\pi_{\theta}(K_{t+1}|\tilde{q}_t)$ and $Q_{\phi}(\tilde{q}_t, K_{t+1})$ respectively, where θ and ϕ are the parameters of the DNNs. To ensure the positiveness of a Lyapunov function, $L_{\phi}(\tilde{q}_t, K_{t+1})$ is selected as the square of a deep neural network as $L_{\phi}(\tilde{q}_t, K_{t+1}) = f_{\phi}^{\top}(\tilde{q}_t, K_{t+1})f_{\phi}(\tilde{q}_t, K_{t+1})$, where f is the vector output of a DNN with parameters ϕ . On the other hand, the stochastic policy $\pi_{\theta}(K_{t+1}|\tilde{q}_t)$ is parametrised by a deep neural network f_{θ} that depends on the state \tilde{q}_t and a Gaussian noise ϵ .

Solving the above constrained optimisation problem is equivalent to minimising the following objective function:

$$J(\theta) = \mathbb{E}_{\tilde{q}_t, a_t, \tilde{q}_{t+1}, c_t \sim D} [\alpha(\ln(\pi_{\theta}(f_{\theta}(\tilde{q}_t, \epsilon) | \tilde{q}_t)) + \mathcal{H}_t) + \lambda (\ln L_{\phi}(\tilde{q}_{t+1}, f_{\theta}(\tilde{q}_{t+1}, \epsilon)) - \ln L(\tilde{q}_t, a_t) + \alpha_2)]$$
(24)

TABLE I Hyperparameters of the proposed estimator

Hyperparameters	Value		
Tryperparameters	value		
Time horizon	1000		
SGD batch size	256		
Actor learning rate	1e-4		
Critic learning rate	3e-4		
Lyapunov learning rate	3e-4		
Target entropy	NAN		
Soft replacement (τ)	5e-3		
Discount (γ)	0.999		
α_3	0.2		
Structure of a_{ϕ}	(128,64,32)		
Structure of L_{θ}	(128,64,32)		

where D is the replay memory of the training samples, α and λ are Lagrange multipliers which control the relative importance of constraints in (23).

In the actor-critic framework, the parameters of policy network are updated through stochastic gradient descent of (24). It can be proved that the policy can converge to an optimal one that ensures the orientation estimate error $\mathbb{E}[\|\tilde{q}_t\|^2]$ converges to a constant as $t \to \infty, \forall \tilde{q}_t \in S$. The proof is standard and omitted due to page limits. The readers can refer to Section IV-D in [25] for more details. Pseudo code of the proposed Lyapunov-based reinforcement learning orientation estimation (LRLOE) algorithm is shown in Algorithm 1.

Algorithm 1 LRLOE algorithm

1: Set the initial parameters ϕ for the Lyapunov function \mathcal{L}_{ϕ} , θ for the estimator gain policy $\pi_{\theta}(K_1|\tilde{q}_0)$, λ for the Lagrangian multiplier, α for the temperature parameter, and the replay memory \mathcal{D}

2: Set the target parameter $\bar{\theta}$ as $\bar{\theta} \leftarrow \theta$

3: while Training do

- 4: **for** each data collection step **do**
- 5: Choose estimator gain K_{k+1} using $\pi_{\theta}(K_{k+1}|\tilde{q}_k)$
- 6: Simulate (1) and (2) with the orientation estimator (3) to collect samples \tilde{q}_k
- 7: $\mathcal{D} \leftarrow \mathcal{D} \cup \tilde{q}_k$
- 8: end for

9: update L_{ϕ} , π_{θ} , λ by optimising (24)

10: end while

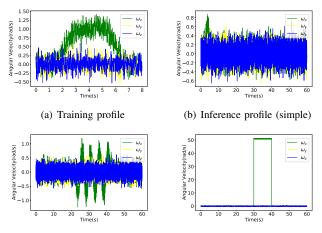
11: Output θ^* , ϕ^* , λ^* , and α^*

V. EXPERIMENTAL RESULTS

In this section, we train and inference on both simulated and real datasets. The estimation results has exhibited good performance, compared with three well-known oirentation estimation algorithms: EKF [5], UKF [10], and complementary filter [6].

A. Results for simulated dataset

The RL policy is trained on a relatively trivial profile (see figure. 2(a)), then tested on three other independent profiles



(c) Inference profile (medium) (d) Inference profile (complicated)

Fig. 2. The angular velocity profiles used for training and inference.

(see figure. 2(b), 2(c) and 2(d)). The sampling rate is set to 100 Hz (consistent with real data in Section V-B). The sensor noise are sampled with the following distribution [3]:

$$e_{\omega,t} \sim \mathcal{N}(0, \Sigma_{\omega}), \qquad \Sigma_{\omega} = 0.0003,$$

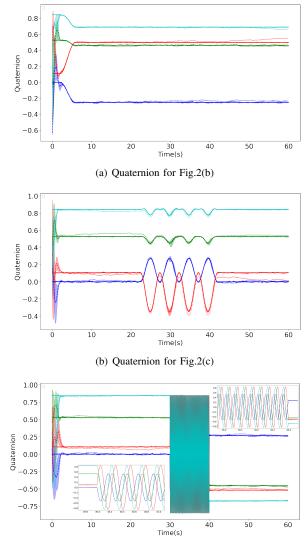
$$e_{a,t} \sim \mathcal{N}(0, \Sigma_{a}), \qquad \Sigma_{a} = 0.0005,$$

$$e_{m,t} \sim \mathcal{N}(0, \Sigma_{m}), \qquad \Sigma_{m} = 0.0003$$
(25)

During inference, the covariance of the measurement noise is increased as Σ_{ω} to 0.03. The hyperparameters of Algorithm 1 are showed in Table I. The last 20% of training data is used for validation. We independently train 20 policies and select the one with the lowest validation error for inference. The initial state estimate is sampled from a normal distribution around the true initial orientation with standard deviation of 0.1. The inference results of our proposed method are shown in Fig. 3, showing accurate estimation for all three different angular velocity profiles.

Furthermore, We compare our proposed DRL-based estimation method with the EKF [5], the UKF [10], and the complementary filter [6]. The root of mean square errors (RMSE) of the decoupled Euler parameters for pitch, ϕ , roll, θ and heading, ψ angles, corresponding to rotations around x, y, z axis respectively is chosen as the estimation performance and 200 Monte Carlo simulations are run for each algorithm. As shown in Fig. 4, the DRL-based estimation method achieves good performance as the EKF and exceeds the UKF and complementary filter under the inference profiles in Fig. 2(c).

The orientation estimation of the EKF, UKF and the complementary filter for a lage angular velocity are shown in Fig. 5. It can be found that the UKF yields a large estimate error under high noise level, and both the EKF and complementary filter perform poor as the estimate error accumulates in the long period of rapid rotation. In comparison, our proposed DRL-based estimation method has learnt error compensation from a large training dataset, thus showing more robust to different movement profiles.



(c) Quaternion for Fig.2(d)

Fig. 3. Quaternion for different angular velocities in Fig. 2(b), 2(c) and 2(d). The solid, dashed and dotted lines correspond to ground truth, measured and estimated quaternion respectively. The shaded areas correspond to the standard deviation over 20 independent runs. For the complicated profile, a zoomed estimation profile for high angular velocity is also showed in Fig. 3(c).

B. Results for real dataset

We apply our algorithm for real dataset from [3] (see Fig. 6). The data is collected from the Trivisio Colibri Wireless IMU [30] with a logging rate of 100Hz. The reference measurement of the orientation is provided as ground truth from a motion capture equipment [31] by tracking the optimal markers fixed to the sensor platform. The optical and IMU data has been time-synchronised and aligned beforehand.

The dataset is 100 seconds long and splited into training and inference dataset separately. The first half of the collected data is used for training and the rest for inference. In the training dataset, we randomly selected with a consecutive sequence of a length of 1000 samples as a training episode. We test in two scenarios (see Fig. 6): (1) inference including the training dataset from t = 0 where the initial estimation $\hat{q}_{t=0s}^{\text{the}}$

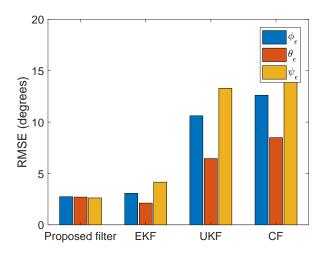
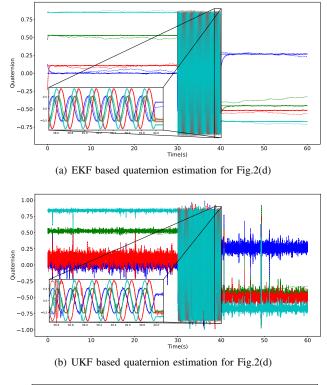


Fig. 4. The RMSE of the proposed filter, Extended Kalman filter (EKF), Unscented Kalman filter (UKF) and Complementary filter (CF) for the inference profiles in Fig. 2(c), 3(b) and 2(d).



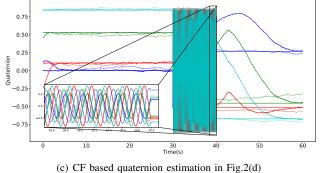


Fig. 5. Comparison with other classic estimation algorithms for large angular velocities in Fig. 2(d).

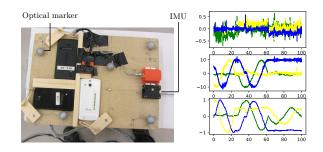


Fig. 6. Real dataset (adapted from Fig. 4.2 and 4.3 in [3]). Left: A snapshot of the platform for collecting real dataset Right: Measurements from an accelerometer $(y_{a,t}, \text{top})$, a gyroscope $(y_{\omega,t}, \text{middle})$ and a magnetometer $(y_{m,t}, \text{bottom})$ for 100 seconds of data collected with the IMU shown in the left figure.).

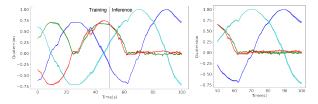


Fig. 7. Quaternion for real data. The solid, dashed and dotted lines correspond to ground truth, measured and estimated quaternion respectively. The left and right figures are corresponding for scenarios 1 and 2 respectively. In the right figure, the dashed line is obtained by averaging over 50 independent trails.

is simply the measurement; (2) inference without training dataset from t = 50s where $\hat{q}_{t=50s}^{\text{nb}}$ is normally distributed around the true initial orientation with a standard deviation of 0.01. 50 independent trials are performed. The estimation results are showed in Fig. 7.

TABLE II RMSE of the orientation estimates

RMSE	Yaw[°]	Pitch[°]	Roll[°]
Proposed algorithm	1.9423	2.1048	0.8353
Extended Kalman Filter	2.0411	1.5272	1.2488
Unscented Kalman Filter	20.1370	20.3494	38.6775
Complementary Filter	1.2015	1.3892	0.8972

In the second scenario, the second half of the dataset is also tested on the EKF, UKF and complementary filter methods. The estimation results are quantified and summarised in Table II. Since the initial estimations and the gyroscope measurements are relatively accurate, results indicate that our proposed algorithm has achieved similar performance with the three state-of-art algorithms.

VI. CONCLUSION

Orientation estimation using inertial sensors combined with magnetometer is well studied and many algorithms have been proposed. However, There hardly exist any algorithms with theoretical guarantees of estimation convergence. In this paper, we propose a reinforcement learning based orientation estimation method and prove that its estimate error converges to a positive scalar in mean square with a guarantee. The proposed method shows superior estimation performance compared with some well-known ones in terms of arbitrary estimation initialisation and adaptation to very large angular velocities.

VII. ACKNOWLEDGEMENT

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